

UNCOUPLED VIBRATIONS IN FUNCTIONALLY GRADED TIMOSHENKO BEAM

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ABSTRACT

Free vibration of FGM Timoshenko beam is investigated on the base of the power law distribution of FGM. Taking into account the actual position of neutral plane enables to obtain general condition for uncoupling of axial and flexural vibrations in FGM beam. This condition defines a class of functionally graded beams for which axial and flexural vibrations are completely uncoupled likely to the homogeneous beams. Natural frequencies and mode shapes of uncoupled flexural vibration of beams from the class are examined in dependence on material parameters and slenderness.

Keywords: FGM, Timoshenko beam; Modal analysis; Coupled vibrations.

1. INTRODUCTION

The functionally graded material (FGM) that usually composes of metal and ceramic constituents has been proved to be an advanced composite compared to the layered ones. The fundamentals of manufacturing technology, modeling and analysis of that material were reviewed in [1 - 2]. Though the powerful methods such as finite element [3 - 5], dynamic stiffness [6] and spectral element [7] have been all developed for analysis of FGM structures, the analytical method has still remained the most efficient tool for dynamic analysis of beam-like FGM structures. Aydogdu and Taskin [8] have examined different high-order shear deformation theories by computing natural frequencies of simply supported functionally graded beam and shown that the classical beam theory gives higher results. Li [9] developed a theory of functionally graded Timoshenko beam neglecting the axial displacement and used to study flexural waves and free vibration of Timoshenko beam. Pradhan and Chakraverty [10] studied natural frequencies of both Euler-Bernoulli and Timoshenko functionally graded beams in dependence on material power-law exponent using Rayleigh-Ritz method. Authors of Ref. [11] investigated effect of slenderness ratio (L/h) and the power-law exponent (n) on natural frequencies of a functionally graded beam using the first-order shear deformation theory of beam. Wei et al. [12] and Aydin [13] studied free vibration of functionally graded beam with edge cracks and dynamic responses of functionally graded beams to moving loads were obtained in [14, 15]. Note that most of the aforementioned theories developed for dynamic analysis of

functionally graded beam are based on the assumption that neutral plane coincides with the mid-plane of beam. This is not true for functionally graded beam, especially, in the case of high gradient of elasticity. Recently, Eltaher et al. [16] have studied effect of exact position of neutral axis on natural frequencies of functionally graded Euler-Bernoulli beam and stated that the mid-plane theory of FGM beam leads natural frequencies to be overestimated. The authors of present paper have investigated material constants calculated for functionally graded Timoshenko beam based on the neutral-plane theory [17] and shown that coupling of axial and flexural vibrations in FGM beam is strongly dependent on the material parameters. Nevertheless, the coupling of axial and flexural vibrations in FGM beam, to the authors' knowledge, has not been thoroughly studied.

Objective of this study is to investigate uncoupled vibrations in functionally graded Timoshenko beam. Namely, the dynamic problem is first formulated for functionally graded Timoshenko beam taking into account the actual position of neutral axis. This enables to obtain general condition for uncoupling of axial and flexural vibration in FGM Timoshenko beams. Numerical analysis of modal parameters of uncoupled flexural vibration is carried out to illustrate and validate the proposed theoretical development.

2. UNCOUPLED VIBRATION CONDITION

Consider a beam of length L , cross-section area $A = b \times h$ made of FGM with the parameters varying accordingly to the power law

$$\begin{Bmatrix} E(z) \\ G(z) \\ \rho(z) \end{Bmatrix} = \begin{Bmatrix} E_b \\ G_b \\ \rho_b \end{Bmatrix} + \begin{Bmatrix} E_t - E_b \\ G_t - G_b \\ \rho_t - \rho_b \end{Bmatrix} \left(\frac{z}{h} + \frac{1}{2} \right)^n, \quad -h/2 \leq z \leq h/2, \quad (2.1)$$

where E , G and ρ stand for elasticity, shear modulus and material density and indexes t and b denote the top and bottom materials; z is ordinate of the point from the central axis at high $h/2$. Assuming linear theory of deformation the displacement fields in the cross-section at x are

$$u(x, z, t) = u_0(x, t) - (z - h_0)\theta(x, t); \quad w(x, z, t) = w_0(x, t), \quad (2.2)$$

with $u_0(x, t)$, $w_0(x, t)$ being the axial and flexural displacements of neutral axis that is located at the high h_0 from the central axis; θ is slope of the cross-section. Therefore, constituting equations get the form

$$\varepsilon_x = \partial u_0 / \partial x - (z - h_0) \partial \theta / \partial x; \quad \gamma_{xz} = \partial w_0 / \partial x - \theta \quad (2.3)$$

and

$$\sigma_x = E(z)\varepsilon_x; \quad \tau_{xz} = \kappa G(z)\gamma_{xz}. \quad (2.4)$$

Using Hamilton principle equations of motion can be derived for free vibration as

$$\begin{aligned} (I_{11}\ddot{u} - A_{11}u'') - (I_{12}\ddot{\theta} - A_{12}\theta'') &= 0; \\ (I_{12}\ddot{u} - A_{12}u'') - (I_{22}\ddot{\theta} - A_{22}\theta'') + A_{33}(w' - \theta) &= 0; \\ I_{11}\ddot{w} - A_{33}(w'' - \theta') &= 0, \end{aligned} \quad (2.5)$$

where

$$\begin{aligned} (A_{11}, A_{12}, A_{22}) &= \int_A E(z)(1, z - h_0, (z - h_0)^2) dA; \quad A_{33} = \kappa \int_A G(z) dA; \\ (I_{11}, I_{12}, I_{22}) &= \int_A \rho(z)(1, z - h_0, (z - h_0)^2) dA. \end{aligned} \quad (2.6)$$

Based on the power law (2.1) for FGM the constants (2.6) can be calculated as follow

$$A_{11} = bhE_b F_1(R_1); \quad A_{12} = bh^2 E_b F_2(R_1); \quad A_{22} = bh^3 E_b F_3(R_1); \quad A_{33} = bh \psi G_b F_1(R_3); \quad (2.7)$$

$$I_{11} = bh\rho_b F_1(R_2); \quad I_{12} = bh^2 \rho_b F_2(R_2); \quad I_{22} = bh^3 \rho_b F_3(R_2);$$

$$F_1(x) = \frac{(x+n)}{(1+n)}; \quad F_2(x) = \frac{2x+n}{2(2+n)} - \frac{x+n}{(1+n)} \alpha; \quad F_3(x) = \frac{3x+n}{3(3+n)} - \frac{2x+n}{(2+n)} \alpha + \frac{x+n}{(1+n)} \alpha^2;$$

$$R_1 = E_t / E_b; \quad R_2 = \rho_t / \rho_b; \quad R_3 = G_t / G_b; \quad \alpha = 1/2 + h_0 / h.$$

It can be seen from Eq. (2.5) that the coefficient A_{12}, I_{12} characterize coupling of axial and flexural vibrations. Indeed, under the condition

$$I_{12} = A_{12} = 0, \quad (2.8)$$

the first equation in (2.5) is uncoupled with two next ones. Note that if neutral axis is assumed to be coincident with the central one, i. e. $\alpha = 1/2$ the uncoupling condition (2.8) leads to either $n = 0$ or $R_1 = R_2 = 1$. This implies that axial and flexural vibrations are uncoupled only for homogeneous beam. However, taking account of exact position of neutral axis determined in [16] as

$$\bar{h}_0 = h_0 / h = \frac{n(R_1 - 1)}{2(n + 2)(n + R_1)}, \quad (2.9)$$

the coefficient $A_{12} = 0$ and

$$\bar{I}_{12} = \frac{I_{12}}{bh^2 \rho_b} = \frac{(R_2 - R_1)n}{2(2+n)(R_1 + n)}. \quad (2.10)$$

Therefore, axial and flexural vibrations may be uncoupled not only for homogeneous beam, when $n = 0$ but also for FGM beam such that

$$R_1 = R_2. \quad (2.11)$$

This type of functionally graded material can be called proportional for which

$$\frac{A_{11}}{I_{11}} = \frac{E_t}{\rho_t} = \frac{A_{22}}{I_{22}} = \frac{E_b}{\rho_b} = C_a^2. \quad (2.12)$$

Under the conditions (2.12) equations of uncoupled axial and flexural vibrations are reduced to

$$\ddot{u} - C_a^2 u'' = 0; \quad (2.13)$$

$$\ddot{\theta} - C_a^2 \theta'' - \Omega_2^2 (w' - \theta) = 0; \quad \ddot{w} - \Omega_1^2 (w'' - \theta') = 0; \quad (2.14)$$

$$\Omega_2^2 = A_{33} / I_{22}; \quad \Omega_1^2 = A_{33} / I_{11}. \quad (2.15)$$

Eq. (2.13) describes axial vibration of an equivalent homogeneous beam with constant wave speed C_a and purely flexural vibration of FGM beam is governed by Eq. (2.14). Note that the purely flexural vibration of a FGM beam was studied in [9] by neglecting axial displacement ($u = 0$), but it is not exactly ($R_1 / R_2 = 1.03$) uncoupled flexural vibration of the proportional functionally graded (PFG) beam what is subject of subsequent sections

3. FREE UNCOUPLED VIBRATION

Since the theory of axial vibration described by Eq. (2.13) for homogenous beam has been well developed, in the present work only uncoupled flexural vibration governed by Eq. (2.14) is investigated. Thus, seeking solution of Eq. (2.14) in the form

$$\theta(x, t) = \Theta(x)e^{i\omega t}; w(x, t) = W(x)e^{i\omega t}, \quad (3.1)$$

one gets

$$(\omega^2 I_{22} \Theta + A_{22} \Theta'') + A_{33} (W' - \Theta) = 0; \omega^2 I_{11} W + A_{33} (W'' - \Theta') = 0. \quad (3.2)$$

Using the following vector $\mathbf{z} = \{\Theta, W\}^T$ and matrices

$$\mathbf{A}_2 = \begin{bmatrix} A_{22} & 0 \\ 0 & A_{33} \end{bmatrix}; \mathbf{A}_1 = \begin{bmatrix} 0 & A_{33} \\ -A_{33} & 0 \end{bmatrix}; \mathbf{A}_0 = \begin{bmatrix} \omega^2 I_{22} - A_{33} & 0 \\ 0 & \omega^2 I_{11} \end{bmatrix}; \quad (3.3)$$

Eq. (3.3) can be rewritten in the matrix form

$$\mathbf{A}_2 \mathbf{z}'' + \mathbf{A}_1 \mathbf{z}' + \mathbf{A}_0 \mathbf{z} = 0. \quad (3.4)$$

Now, seeking solution of Eq. (3.4) in the form $\mathbf{z}_0 = \mathbf{d}e^{\lambda x}$ leads the equation to

$$[\lambda^2 \mathbf{A}_2 + \lambda \mathbf{A}_1 + \mathbf{A}_0] \mathbf{d} = 0. \quad (3.5)$$

The latter equation would have nontrivial solution with respect to constant vector \mathbf{d} under the condition

$$\det[\lambda^2 \mathbf{A}_2 + \lambda \mathbf{A}_1 + \mathbf{A}_0] = 0, \quad (3.6)$$

that can be expressed in the form

$$\lambda^4 + a\lambda^2 + b = 0, \quad (3.7)$$

where

$$a = \omega^2 (I_{11} / A_{33} + I_{22} / A_{22}); b = (\omega^2 I_{11} / A_{33})(\omega^2 I_{22} / A_{22} - A_{33} / A_{22}). \quad (3.8)$$

In general, equation (3.7) is elementarily solved and gives in result

$$\lambda_{1,2}^2 = (-a \pm \sqrt{a^2 - 4b}) / 2 = \eta_{1,2}. \quad (3.9)$$

Note first that Eq. (3.7) would have trivial root ($\lambda = 0$) under the condition

$$\omega = \omega_c = \sqrt{A_{33} / I_{22}} = \Omega_2, \quad (3.10)$$

termed as cutoff frequency of the beam. Otherwise, the Eq. (3.7) has all four imaginary roots for $\omega > \omega_c$ and two real roots if $\omega < \omega_c$. Hence, four roots of Eq. (3.7) are

$$\lambda_{1,3} = \pm k_1; \lambda_{2,4} = \pm k_2; k_j = \sqrt{\eta_j}, j = 1, 2 \quad (3.11)$$

and general solution of Eq. (3.4) can be represented as

$$\mathbf{z} = \begin{Bmatrix} \Theta \\ W \end{Bmatrix} = \begin{Bmatrix} d_{11}e^{k_1x} + d_{12}e^{k_2x} + d_{13}e^{-k_1x} + d_{14}e^{-k_2x} \\ d_{21}e^{k_1x} + d_{22}e^{k_2x} + d_{23}e^{-k_1x} + d_{24}e^{-k_2x} \end{Bmatrix}. \quad (3.12)$$

Taking into account the second equation in (3.2) one gets

$$d_{21} = \alpha_1 d_{11}, d_{22} = \alpha_2 d_{12}, d_{23} = -\alpha_1 d_{13}, d_{24} = -\alpha_2 d_{14}, \quad (3.13)$$

where

$$\alpha_1 = k_1 A_{33} / (\omega^2 I_{11} + k_1^2 A_{33}), \alpha_2 = k_2 A_{33} / (\omega^2 I_{11} + k_2^2 A_{33}). \quad (3.14)$$

Therefore, expression (3.12) can be now rewritten in the form

$$\mathbf{z}(x, \omega) = \mathbf{G}(x, \omega) \mathbf{d}, \quad (3.15)$$

with $\mathbf{d} = (d_1, \dots, d_4)^T = (d_{11}, \dots, d_{14})^T$ and

$$\mathbf{G}(x, \omega) = [\mathbf{G}_1(x, \omega) \quad \mathbf{G}_2(x, \omega)];$$

$$\mathbf{G}_1(x, \omega) = \begin{bmatrix} e^{k_1x} & e^{k_2x} \\ \alpha_1 e^{k_1x} & \alpha_2 e^{k_2x} \end{bmatrix}; \mathbf{G}_2(x, \omega) = \begin{bmatrix} e^{-k_1x} & e^{-k_2x} \\ -\alpha_1 e^{-k_1x} & -\alpha_2 e^{-k_2x} \end{bmatrix}. \quad (3.16)$$

The solution (3.15) should fulfill conditions at the ends of the beam that can be represented in the form

$$\mathbf{B}_0 \{\mathbf{z}\}_{x=0} = 0; \mathbf{B}_L \{\mathbf{z}\}_{x=L} = 0, \quad (3.17)$$

where $\mathbf{B}_0, \mathbf{B}_L$ are differential operators of dimension 2x2. For instance, the operators $\mathbf{B}_0, \mathbf{B}_L$ for conventional boundary conditions for respectively simple support; clamp and free end would be

$$\mathbf{B}_s = \begin{bmatrix} -A_{22}\partial_x & 0 \\ 0 & 1 \end{bmatrix}; \mathbf{B}_c = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix}; \mathbf{B}_e = \begin{bmatrix} -A_{22}\partial_x & 0 \\ -A_{33} & A_{33}\partial_x \end{bmatrix}.$$

Now, decomposing the vector $\mathbf{d} = \{\mathbf{d}_1, \mathbf{d}_2\}^T$ with $\mathbf{d}_1 = \{d_1, d_2\}^T; \mathbf{d}_2 = \{d_3, d_4\}^T$ the condition at the left end of beam can be expressed as

$$\mathbf{B}_{01} \mathbf{d}_1 + \mathbf{B}_{02} \mathbf{d}_2 = 0, \quad (3.18)$$

where

$$\mathbf{B}_{01}(\omega) = \mathbf{B}_0 \{\mathbf{G}_1(x, \omega)\}_{x=0}; \mathbf{B}_{02}(\omega) = \mathbf{B}_0 \{\mathbf{G}_2(x, \omega)\}_{x=0}.$$

Obviously, Eq. (3.18) allows eliminating one of the vectors $\mathbf{d}_1, \mathbf{d}_2$ and as result one is able to reconstruct the solution $\mathbf{z}(x)$ as

$$\mathbf{z}(x, \omega) = \mathbf{G}_0(x, \omega) \mathbf{D} \quad (3.19)$$

with $\mathbf{G}_0(x, \omega)$ being 2x2 dimension matrix function and arbitrary constant vector $\mathbf{D} = \{D_1, D_2\}^T$. Applying boundary conditions at the other end ($x = L$) of beam for solution (3.19) one gets

$$[\mathbf{G}_L(\omega)]\{\mathbf{D}\} = 0; \tag{3.20}$$

$$\mathbf{G}_L(\omega) = \mathbf{B}_L \{ \mathbf{G}_0(x, \omega) \}_{x=L}. \tag{3.21}$$

This equation has nontrivial solution only under the condition

$$L_0(\omega) = \det[\mathbf{G}_L(\omega)] = 0, \tag{3.22}$$

that provides the so-called frequency equation for FGM beam. Each root ω_j of the frequency equation is related to a mode shape

$$\Phi_j(x) = C_j \mathbf{G}_0(x, \omega_j) \bar{\mathbf{D}}_j, \tag{3.23}$$

where C_j is an arbitrary constant and $\bar{\mathbf{D}}_j$ is the normalized solution of (3.18) with $\omega = \omega_j$.

4. RESULTS AND DISCUSSIONS

In this section numerical analysis is examined for PFG beam with the initial material parameters [9]: Steel: $E_b = 210\text{GPa}$, $\rho_b = 7850\text{kg/m}^3$, $\mu_1 = 0.3$ (bottom surface) and shear module is calculated as $G = E/2(1 + \mu)$. The material parameters of the top surface are calculated from those of bottom with proportional ratio r : $E_t = rE_b$; $\rho_t = r\rho_b$. Under the analysis the dimensionless natural frequencies $\bar{\omega}_j = (\omega_j L^2 / h) \sqrt{\rho_b / E_b}$ and related mode shapes are examined in dependence on the proportional ratio; exponent of the power law, n (called here material distribution index) and the shear modulus ratios $\gamma = G_t / G_b = r(\mu_2 + 1) / (\mu_1 + 1)$.

First, natural frequencies of homogenous Timoshenko simply supported beam are computed and compared to those obtained in Ref. [6] by the dynamic stiffness method and in Ref. [18] by analytical method. It can be observed excellent agreement of the results that consequently verify the proposed above theory.

Table 1. Comparison of natural frequencies for homogeneous Timoshenko beam.

Freq. No	L/h =10			L/h =30			L/h =100	
	Present	Ref. [18]	Ref. [6]	Present	Ref. [18]	Ref. [6]	Present	Ref. [18]
1	2.8020	2.8020	2.8023	2.8438	2.8438	2.8439	2.8486	2.8486
2	10.6948	10.6947	-	11.3116	11.3116	-	11.3887	11.3887
3	15.6092	15.7080	-	25.2184	25.2184	-	25.6046	25.6046
E =70 GPa, ρ=2700kg/m ³ , μ=0.3								

4.1. Cutoff frequency analysis

Cutoff frequency (3.10) computed as function of proportional ratio in various material distribution index and slenderness is shown in Figs. 1-2. It can be observed from the Figures that cutoff frequency is monotonically increasing with proportional ratio larger 1 for $n \leq 5$ and it is decreasing with growing r for $n > 5$. Obviously, cutoff frequency of homogenous beam is constant and it is rapidly increasing with beam slenderness L/h. Moreover, for a fixed material

distribution index cutoff frequency is weakly dependent on the proportional ratio; the dependence is almost linear for $n = 50$ and strongly nonlinear for $n = 2$.

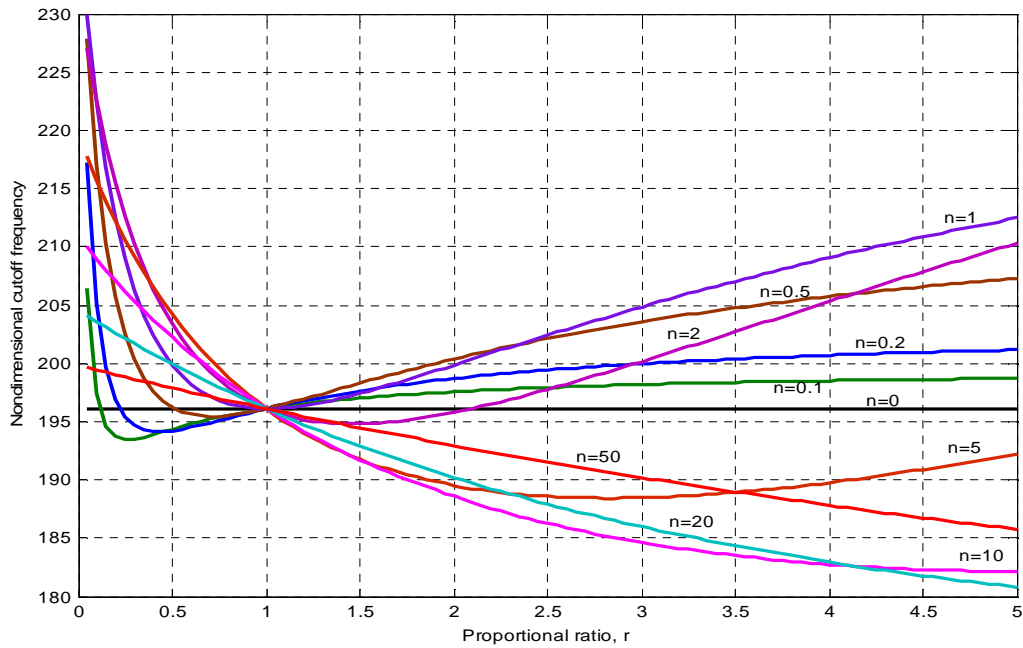


Figure 1. The uncoupled cutoff frequency vs. proportional ratio for various material distribution index n .

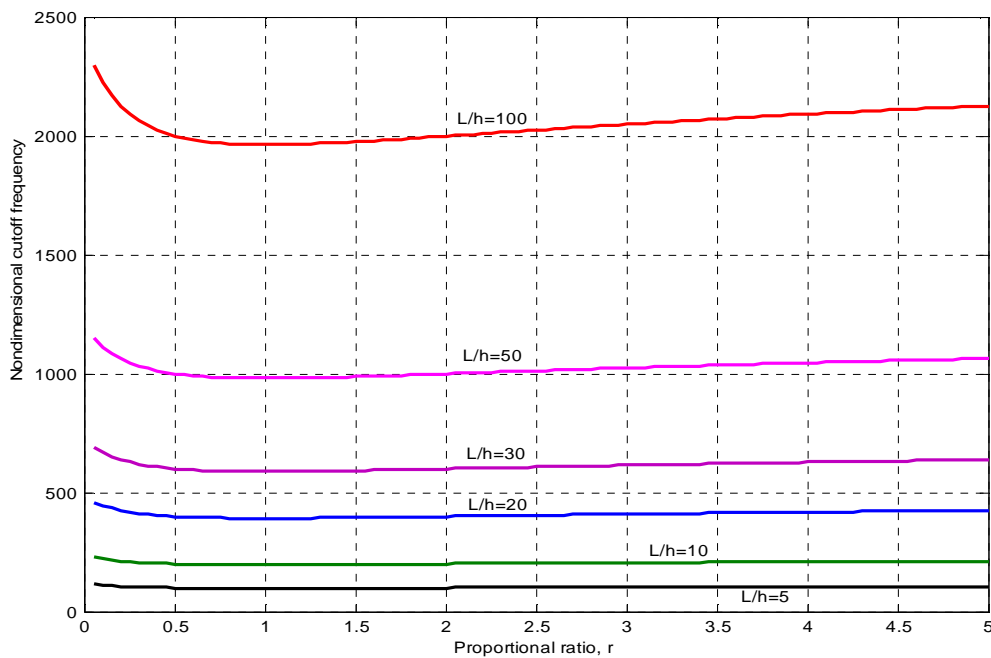


Figure 2. The uncoupled cutoff frequency vs. proportional ratio for various slenderness L/h .

4.2. Natural frequency analysis

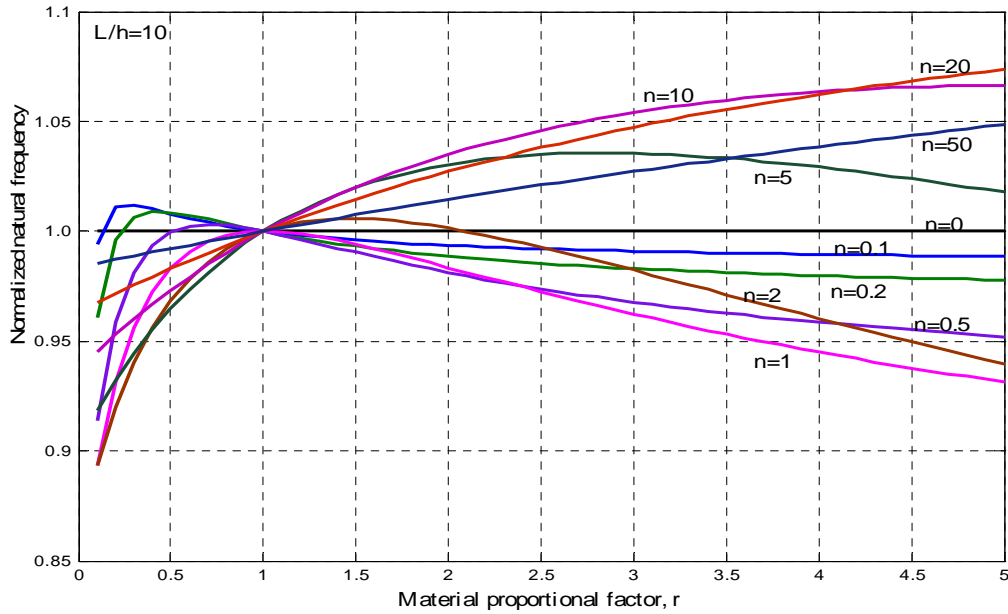


Figure 3. Normalized uncoupled flexural frequency of FGM beam in dependence on the proportional factor r in various material distribution index n .

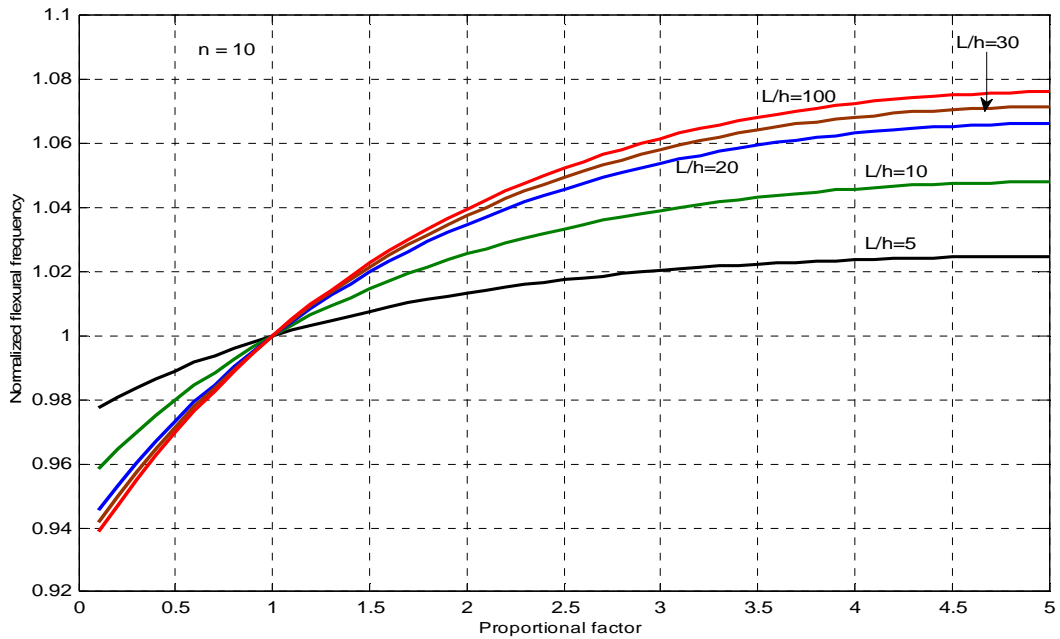


Figure 4. Normalized uncoupled flexural frequency of FGM beam in dependence on the proportional factor R in various slenderness L/h .

Typical variation of natural frequencies versus proportional ratio is shown in Figs. 3-4, in which there are presented natural frequencies normalized by those of homogeneous beam ($r = 1$). It can be seen that flexural natural frequencies of PFG beam are limited to the range (0.9-1.1) times of the frequencies of homogeneous beam. They are less than those of homogeneous beam when $n < 5$ and become greater for $n \geq 5$. The linear dependence of

natural frequencies on proportional ratio gets to be if $n = 50$. Normalized natural frequencies are all monotonically increasing with slenderness for $r > 1$ and decreasing for $r < 1$.

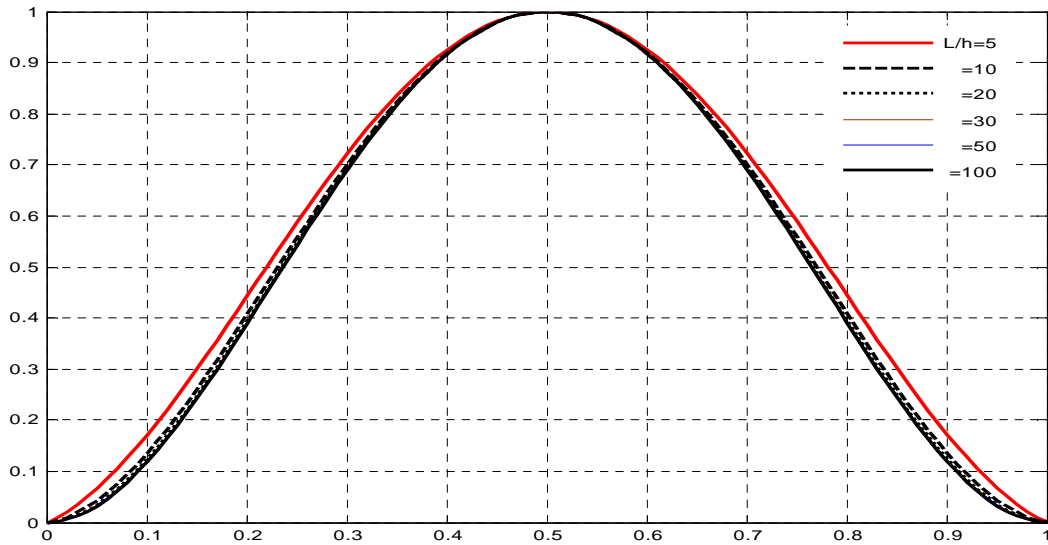


Figure 5. First mode shape in dependence on the slenderness L/h .

4.3. Mode shape analysis

Numerical analysis shows that mode shapes of the PFG beam are not affected by material parameters such as proportional ratio r and material distribution index n . They may be slightly modified by various slenderness of the beam what is demonstrated in Figs. 5-7 where first three mode shapes of clamped functionally graded Timoshenko beam are presented.

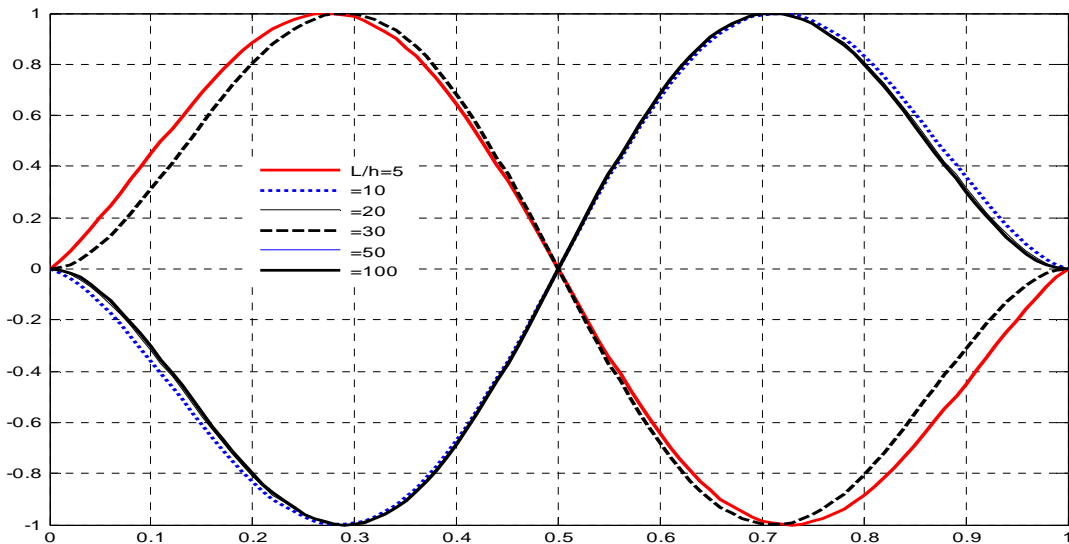


Figure 6. Second mode shape in dependence on the slenderness L/h .

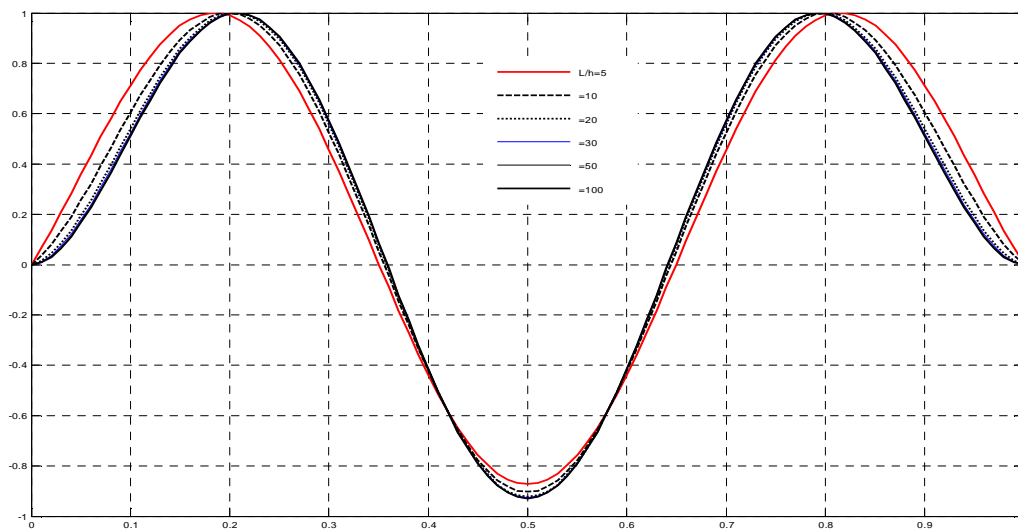


Figure 7. Third mode shape in dependence on the slenderness L/h .

5. CONCLUSION

The main results obtained in present study are as following:

1. In the framework of the proposed theory of vibration for functionally graded beam based on true position of neutral plane, a condition for uncoupling of axial and flexural vibration modes has been obtained.
2. It was shown that uncoupled axial vibration of such the beam remains completely similar to that of homogeneous beam.
3. Numerical analysis has demonstrated that natural frequencies including the cutoff frequency in uncoupled flexural vibration are typically dependent on the material and geometrical parameters of the beam.
4. Mode shapes of the uncoupled flexural vibration are insensitive to material parameters; they are dependent only on slenderness of the functionally graded beam.
5. All the above mentioned concluding remarks provide useful instructions for modal testing and identification of functionally graded beam. Moreover, the obtained natural frequencies of an FGM beam in dependence on the material properties enable one to control not only the vibration characteristics but also the stiffness and material density of the beam.

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TÓM TẮT

DAO ĐỘNG TÁCH RỜI CỦA DÀM TIMOSHENKO CÓ CƠ LÝ TÍNH BIẾN THIÊN

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Bài báo này nghiên cứu dao động riêng của dầm Timoshenko có cơ lý tính biến đổi theo quy luật lũy thừa. Việc tính đến vị trí thực của trục trung hòa (không phải là trục giữa dầm) cho phép ta nhận được điều kiện để dao động dọc trục và dao động uốn tách rời nhau (trở nên độc lập) giống như dầm đồng nhất. Tuy nhiên các tham số của các dạng dao động tách rời đó vẫn là của dầm có cơ lý tính biến thiên (chứ không phải của dầm đồng nhất). Ở đây, nghiên cứu tần số và dạng riêng trong dao động uốn tách rời của dầm FGM phụ thuộc vào các tham số vật liệu và hình học.

Từ khóa: Vật liệu cơ lý tính biến thiên, dầm Timoshenko, phân tích dao động, dao động quan liên